



BOMBARDIER CHALLENGER 300



A true executive office for the skies, it is the definitive, super-midsize corporate jet, and the first clean sheet design to satisfy customer needs unmet by any other aircraft. A sophisticated and extremely comfortable transcontinental high-performance jet, the rugged and refined Bombardier Challenger^{*} 300 was built to provide outstanding reliability and maintain its unsurpassed value over time.

- Best performance and value in its class
- 'Market-designed' cabin functionality and flexibility
- Supreme reliability, built for long-term service

GENERAL

Capacity

Crew: 2

Typical seating: 8 to 9 (Double club configuration)

Engines

Honeywell HTF7000 turbofans Thrust: 6,826 lb (30.4 kN)

Flat rated to: ISA + 15°C (30°C / 86°F)

Avionics

- Rockwell Collins Pro Line 21 system with four LCD displays
- Engine Instrument/Crew Alerting System (EICAS)
- Maintenance Diagnostics Computer
- · Enhanced Ground Proximity Warning System (EGPWS)
- Traffic Collision Avoidance System (TCAS II)
- Emergency Locator Transmitter (ELT)

PERFORMANCE

Range

Maximum range: 3,100 NM 3,568 SM 5,741 km (Standard BOW, NBAA IFR Reserves, ISA, with 8 pax/2 crew and maximum allowable fuel)

Speed	Mach	kt	mph	km/h
High-speed	0.82	470	541	870
Typical cruise speec	0.80	459	528	850

Airfield Performance

Takeoff distance (SL, ISA, MTOW): 4,810 ft (1,466 m)

Landing distance (SL, ISA, MLW): 2,600 ft (792 m)

Operating Altitude

Maximum operating altitude: 45,000 ft (13,716 m)

Initial cruise altitude (MGTOW): 41,000 ft (12,496 m)

DIMENSIONS

Exterior

Length: 68.63 ft (20.92 m) Wingspan: 63.84 ft (19.46 m) Wing area: 522 ft² (48.50 m²) (basic)

Height overall: 20.33 ft (6.20 m)

Interior

Cabin length: 28.6 ft (8.72 m) (from cockpit divider to end of pressurized compartment)

Cabin width: 7.17 ft (2.19 m) (centerline)

Cabin width: 5.08 ft (1.55 m) (floorline)

Cabin height: 6.08 ft (1.85 m)

Floor area: 146 ft² (13.56 m²) (excluding cockpit)

Cabin volume: 860 ft³ (24.35 m³) (excluding cockpit and baggage)

Weights

- A. Maximum ramp weight: 39,000 lb (17,690 kg)
- B. Maximum takeoff weight: 38,850 lb (17,622 kg)
- C. Maximum landing weight: 33,750 lb (15,309 kg)
- D. Maximum zero fuel weight: 27,000 lb (12,247 kg)
- E. Standard basic operating weight^A: 23,500 lb (10,659 kg)
- F. Maximum fuel weight:
- 14,150 lb (6,418 kg)

Maximum payload (D-E): 3,500 lb (1,588 kg)

Payload - maximum fuel (A-E-F): 1,350 lb (612 kg)

Fuel with maximum payload: 12,000 lb (5,443 kg)

△ Includes interior, paint, 2 crew + operating items.

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Performance is based on maximum takeoff/landing weight, sea level, standard day (ISA) conditions, unless otherwise noted. All data in this document is approximate, for discussion purposes only and may be subject to certain conditions. This document does not constitute an offer, commitment, representation or warranty of any kind. All data is subject to change without prior notice. The interior images shown are for information purposes only and may represent some optional configurations. The final configuration and performance of the aircraft shall be as per the contractually agreed upon purchase agreement.

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